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#### Flexural strengthening

Book Composite for Construction, L. C. Bank, Chapter 9



#### Initial situation prior to strengthening

The effect of the initial load prior to strengthening should be considered in the calculation of strengthened member. Based on the theory of elasticity and with M<sub>0</sub> the service moment (*no* load safety factors are applied) acting on the critical RC section during strengthening, the strain distribution of the member can be evaluated. As M<sub>0</sub> is typically larger than the cracking moment  $M_{cr}$ , the calculation is based on a cracked section.



Externally Bonded FRP: Flexural Fibre Composites, FS24 Masoud Motavalli If M0 is smaller than Mcr, its influence on the calculation of the strengthened member may easily be neglected.

Based on the transformed cracked section, the neutral axis depth  $x_0$ can be solved from:

$$
\frac{1}{2}bx_0^2 + (\alpha_s - 1)A_{s2}(x_0 - d_2) = \alpha_s A_{s1}(d - x_0)
$$

Where:

$$
\alpha_s = \frac{E_s}{E_c}
$$
  
Externally Bonded FRP: Flexural  
Five Composites, FS24  
Masoud Motavalli  
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The concrete strain at the top fiber can be expressed as:

$$
\varepsilon_{c0} = \frac{M_0 x_0}{E_c I_{02}}
$$

Where  $I_{02}$  is the moment of inertia of the transformed cracked section:

$$
I_{02} = \frac{bx_0^3}{3} + (\alpha_s - 1)A_{s2}(x_0 - d_2)^2 + \alpha_s A_{s1}(d - x_0)^2
$$

Bonded FRP: Flexural<br>Externally Bonded FRP: Flexural Fibre Composites, FS24 Masoud Motavalli<br>
Externally Bonded FRP: Flexural Fibre Composites, FS24 Masoud Motavalli<br>
Externally Bonded FRP: Flexural Fibre Composites, FS24 Based on strain compatibility, the concrete strain at the extreme tension fiber can be derived as:

$$
\varepsilon_0 = \varepsilon_{c0} \frac{h - x_0}{x_0}
$$

This strain equals the initial axis strain at the level of the FRP, needed for the evaluation of the strengthened member.

#### Analysis of Ultimate Limit State (ULS)

#### Full composite action Steel yielding followed by concrete crushing



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Calculation of neutral axis depth, x:

$$
0.85.\psi.f_{cd}bx+A_{s2}E_s\varepsilon_{s2}=A_{s1}f_{yd}+A_fE_{fu}\varepsilon_f
$$
  
Where:

$$
\psi=0.8
$$

and:

and:  
\n
$$
\varepsilon_{s2} = \varepsilon_{cu} \frac{x - d_2}{x}
$$
\n( $E_s \varepsilon_{s2}$  not to exceed  $f_{yd}$ )  
\n
$$
\varepsilon_f = \varepsilon_{cu} \frac{h - x}{x} - \varepsilon_0
$$
\nExternally Bonded FRP: Flexural

\nFive Composites, FS24

\nMasoud Motavalli

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$$
(\mathsf{E}_{\mathsf{s}} \varepsilon_{\mathsf{s}2} \text{ not to exceed } \mathsf{f}_{\mathsf{yd}})
$$

Design bending moment capacity:

$$
M_{Rd} = A_{s1} f_{yd} (d - \delta_G x) + A_f E_f \varepsilon_f (h - \delta_G x) + A_{s2} E_s \varepsilon_{s2} (\delta_G x - d_2)
$$

#### Where:

 $\mathcal{S}_G=0.4$ <br>Externally Bonded FRP: Flexural Fibre Composites, FS24 Masoud Motavalli  $\delta_G = 0.4$ 

#### Check if

a) Yielding of tensile steel reinforcement:

$$
\varepsilon_{s1} = \varepsilon_{cu} \frac{d - x}{x} \ge \frac{f_{yd}}{E_s}
$$

b) Straining of the FRP is limited to the ultimate strain:

b) Straining of the FRP is limited to the ultimate strain:  
\n
$$
\mathcal{E}_f = \mathcal{E}_{cu} \frac{h - x}{x} - \mathcal{E}_0 \leq \mathcal{E}_{fid}
$$
\nExtemally Bonded FRP: Flexural

\nFive Composites, FS24

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#### Tee Beams



#### Neutral axis in flange: treat as rectangular section

#### Neutral axis in web: treat as tee section

#### Debonding and bond failure modes

- **Debonding in the concrete near the surface or along a** weakened layer, e.g. along the line of the embedded steel reinforcement.
- **Debonding in the adhesive (cohesion failure).**
- Debonding at the interfaces between concrete and adhesive<br>or adhesive and FRP (adhesion failure).<br>Debonding inside the FRP (interlaminar shear failure).<br>Externally Bonded FRP: Flexural Fibre Composites, FS24 Masoud Motaval **Debonding at the interfaces between concrete and adhesive** or adhesive and FRP (adhesion failure).
- **Debonding inside the FRP (interlaminar shear failure).**



### Debonding Video Clips Non Prestressed CFRP Prestressed CFRP



#### Lap Shear Test













Pull-off Test No. 3









#### Bond failure of RC members strengthened with FRP:

See next lecture given by Dr. Christoph Czaderski



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#### 3. Debonding at flexural cracks

$$
\varepsilon_{\rm f} \leq \varepsilon_{\rm f,lim,d} = 8\% \text{,}
$$

 $\vert f \vert \vert < \vert \vert \frac{\angle I \Gamma_f} {\perp} \vert$  $\overline{x}$   $\Big)$   $\Big\{$   $\overline{\Delta x}$   $\Big\}$  $\overline{\Delta x}$   $\Big)$   $\cong$   $\Big( \overline{\Delta x}$ 

 $\vert \mathsf{F}_{\mathrm{fcr}} \leq \mathsf{F}_{\mathrm{b,R}}$ 

1. End strip debonding failure at the last crack

Summary of the three Swiss Code (SIA 166) verifications

2. Debonding at strong strain increase in strip







4-Point Bendie test, RC beam<br>Istrengthened with a CFRP Strip



#### Serviceability Limit State (SLS)

- **linear elastic material behavior**
- **Cracked section analysis**



Calculation of neutral axis  $x_e$ :

$$
\frac{1}{2}bx_e^2 + (\alpha_s - 1)A_{s2}(x_e - d_2) = \alpha_s A_{s1}(d - x_e) + \alpha_f A_f \left[ h - (1 + \frac{\varepsilon_0}{\varepsilon_c})x_e \right]
$$
  
\nWhere:  
\n
$$
\alpha_f = \frac{E_f}{E_c}
$$
\nAnd the cracking moment for rectangular beams:  
\n
$$
M_{cr} \approx f_{ctrm} \cdot \frac{bh^2}{6}
$$
\nExternally Bonded FRP: Flexural  
\nFibre Composites, FS24  
\n<sup>28</sup>

And the cracking moment for rectangular beams:

$$
M_{cr} \approx f_{ctm} \cdot \frac{bh^2}{6}
$$

#### Stress limitation

limit stresses in the concrete, steel and FRP to prevent

- **damage or excessive creep of the concrete**
- **steel yielding**
- Externally Bonded FRP: Flexural Fibre Composites, FS24 Masoud Motavalli<br>Externally Bonded FRP: Flexural Fibre Composites, FS24 Masoud Motavalli **Excessive creep or creep rupture of the FRP**

 $\sigma_c \leq 0.60 f_{ck}$  under the rare load combination

 $\sigma_c \leq 0.45 f_{ck}$  under the quasi-permanent load combination

$$
D_c \geq 0.4 J_{ck}
$$
 under the quasi-permanent load combination  
where: 
$$
\sigma_c = E_c \mathcal{E}_c
$$
  
Extemally Bonded FRP: Flexural  
Five Composites, FS24  
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To prevent yielding of the steel at service load:

Externally Bonded FRP: Flexural Fibre Composites, FS24 Masoud Motavalli yk e e s s c f x d x <sup>E</sup> . . 0.80 rare load combination

FRP stress under service load should be limited as:

fk e e  $f = E_f$ . $(\varepsilon_c \cdot \frac{\hbar^2}{\hbar^2} - \varepsilon_0) \leq \eta.f$  $\overline{X}_{\epsilon}$  $h - x$  $\sigma_f = E_f . (\varepsilon_c . \frac{E}{m} - \varepsilon_0) \leq \eta.$  $\overline{\phantom{0}}$  $\tilde{v} = {E}_f.({\varepsilon}_c.\frac{n - \varepsilon_e}{\sqrt{2}} - {\varepsilon}_0) \leq \eta.f_{\varepsilon_e}$  quasi-permanent load combination

Where 
$$
\eta = \begin{cases} 0.8 : CFRP \\ 0.5 : AFRP \\ 0.3 : GFRP \end{cases}
$$
  
Externally Bonded FRP: Flexural  
Five Composites, FS24  
Massoud Motavalli  
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#### Verification of deflections

The mean deflection, a, is calculated from:

$$
a = a_1.(1 - \zeta_b) + a_2.\zeta_b
$$

Where  $\mathsf{a}_1$  and  $\mathsf{a}_2$  are the deflections in the uncracked and the fully cracked state, respectively and the distribution coefficient is:

Where 
$$
a_1
$$
 and  $a_2$  are the deflections in the uncracked and the fully cracked state, respectively and the distribution coefficient is:  
\n
$$
\zeta_b = 0 \dots M_k < M_{cr}
$$
\n
$$
\zeta_b = 1 - \beta_1 \beta_2 \cdot \left(\frac{M_{cr}}{M_k}\right)^{n/2} \dots M_k > M_{cr}
$$
\nExtemally Bonded FRP: Flexural  
\nFibre Composites, FS24  
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\n34  

• Where β1 is a coefficient taking into account the bond characteristics of the reinforcement and equals 0.5 and 1 for smooth and deformed steel, respectively;

- $\overline{B}$  β2 is a coefficient taking into account the loading type and equals 0.5 and 1 for long-term and short term loading, respectively.
- Externally Bonded FRP: Flexural Fibre Composites, FS24 Masoud Motavalli • The power n equals 2. For high strength concrete more accuracy is obtained with n equal to 3.

Fibre Composites, FS24 Masoud Motavalli<br>
Externally Bonded FRP: Flexural Fibre Composites, FS24 Masoud Motavalli<br>
Externally Bonded FRP: Flexural Fibre Composites, FS24 Masoud Motavalli The deflection in the uncracked state, a1, and in the fully cracked state, a2, can be calculated by classical elasticity analysis, referring to a flexural stiffness in the uncracked state  $E_{c}I_{1}$  and in the fully cracked state E<sub>c</sub>l<sub>2</sub>, respectively.

#### Verification of crack widths

Neglecting the tension stiffening effect ( $\zeta$  = 1) and assuming  $\varepsilon_0 \approx 0$ 

$$
w_k = 2.1 \rho_{c,eff} \cdot \frac{M_k}{E_s d\rho_{eq}} \cdot \frac{1}{(u_s + 0.694u_f)}
$$

Where the ratio of the effective area in tension is:

$$
\rho_{c, \text{eff}} = \frac{A_{c, \text{eff}}}{bd}
$$

Where the ratio of the effective area in tension is:<br>  $\rho_{c,eff} = \frac{A_{c,eff}}{bd}$ <br>  $\rho_{eq}$  is the equivalent reinforcement ratio and  $u_s$  and  $u_f$  is the bond<br>
perimeter of the steel and FRP reinforcement.<br>
Externally Bonded F  $\mathcal{P}_{eq}$  is the equivalent reinforcement ratio and  $\mathsf{u}_{\mathsf{s}}$  and  $\mathsf{u}_{\mathsf{f}}$  is the bond perimeter of the steel and FRP reinforcement.

#### Summary of design procedure:

- Before strengthening: check ULS and SLS (just to compare with the strengthened member!).
- From the service moment M<sub>0</sub> prior to strengthening determine  $\varepsilon_0$  at the extreme tension fiber.
- Assume full composite action and from the design moment after strengthening<br>determine the required FRP cross section to fulfill the ULS. Verify the ductility<br>requirements.<br>Calculate the deflections in the SLS. If allowable **• Assume full composite action and from the design moment after strengthening** determine the required FRP cross section to fulfill the ULS. Verify the ductility requirements.
- Calculate the deflections in the SLS. If allowable deflection is exceeded, determine the required FRP cross section.
- Calculate the stresses in the concrete, steel and FRP and verify the allowable stresses.
- Verify that the provided FRP bond width is sufficient to control crack widths in the SLS. Increase the FRP width, if necessary, or, given a maximum width, increase the amount (thickness) of FRP.
- Verify the resisting shear force at which bond failure due to shear cracks occurs (ULS).
- Verify the resisting shear force at which bond failure due to shear cracks occurs (ULS).<br>
 Verify that bond failure at the anchorage does not occur. Otherwise mechanical anchorage should be provided.<br>
Externally Bonded Verify that bond failure at the anchorage does not occur. Otherwise mechanical anchorage should be provided.
- Verify that FRP end shear failure is avoided. Provide shear strengthening at the ends if required.
- Verify the accidental situation.
- Verify the shear design resistance of the strengthened member. If needed shear strengthening should be provided.

## Strengthening of a Large Scale Pre-Stressed<br>Bridge Girder Using Carbon Fibre Reinforced<br>Polymers: Strengthening of a Large Scale Pre-Stressed<br>Bridge Girder Using Carbon Fibre Reinforced<br>Polymers:<br>Comparision between Non Prestressed and Polymers: Strengthening of a Large Scale Pre-Stressed<br>Bridge Girder Using Carbon Fibre Reinforced<br>Polymers:<br>Comparision between Non Prestressed and<br>Prestressed CFRP Plates Strengthening of a Large Scale Pre-<br>Bridge Girder Using Carbon Fibre Re<br>Polymers:<br>Comparision between Non Prestress<br>Prestressed CFRP Plates

# Polymers:<br>Comparision between Non Prestressed and<br>Prestressed CFRP Plates<br>Externally Bonded FRP: Flexural Fibre Composites, FS24 Masoud Motavalli



#### Bridge "Viadotto delle Cantine a Capolago"





#### Bridge "Viadotto delle Cantine a Capolago"







#### **Overview**

#### **Reference beam**

- **Beam strengthened with non prestressed CFRP plates** 
	- 6 Sika CarboDur 512 plates, each 15.5 m long
- Beam strengthened with *pre*stressed CFRP plates<br>■ the same type and number of plates<br>■ each plate prestressed approx. 1000 MPa (60 kN)<br>■ anchorage: Empa gradient method<br><br>Externally Bonded FRP: Flexural Fibre Composites **Beam strengthened with prestressed CFRP plates** 
	- $\blacksquare$  the same type and number of plates
	- each plate prestressed approx. 1000 MPa (60 kN)
	- anchorage: Empa gradient method

#### Strengthened with non prestressed CFRP plates













#### Shear stress from Deformeter-measurement



#### Behavior during loading











#### Summary of the three SIA 166 verifications

See next lecture given by Dr. Christoph Czaderski

- 1. End strip debonding failure at the last crack
- 2. Debonding at strong strain increase in strip



 $\vert \mathsf{F}_{_{\sf fcr}} \leq \mathsf{F}_{_{\sf b,R}}$ 

3. Debonding at flexural cracks

$$
\boxed{\varepsilon_{\text{f}} \leq \varepsilon_{\text{f,lim,d}} = 8\%}
$$



#### SIA166 "Externally bonded reinforcement"



$$
\tau_{1,lim} = 2.5 \cdot \tau_c = 2.5 \cdot 2.0 = 5.0 MPa
$$

#### Strengthened with prestressed CFRP plates



#### CFRP plates prestressed approx. 1000 MPa (60 kN)

#### Prestressing using Gradient-method





List of Symbols" <Flexural strongthening>  $M_{\odot}$ : revisice moment  $Mcr$ : cracking moment<br>staal com-rection at (tensile reinforcement)  $A_{54}$ steel cross-rection (Compression reinforcement)  $As_{2}$ Position of the neutral axis prior to strengthening  $x_{o}$  : cross-section width  $6:$ a debth i h = d+ dx Es {Es: steel E-modulus  $\alpha_s$  : Io 2: moment of interpretation of the transformed conclude exchange Ly : bond length  $\sqrt{2}$  $stip$   $(\equiv s)$  $\begin{array}{lllllllll} \mathcal{A} & = & 0.5 & \text{if} & \text{if } & \text{$ 

" List of Symbols" < Flexural strengthening> : maximum FRP force, which can be auchored N<sub>fa</sub>, max b, max: maximum anchorage length ferm : mean concrete tensile strength : FRP tickness  $\epsilon$   $\epsilon$ position of neutral axis at SLS  $X_{e}$  $665 < 2.5$  for TRP stress under sensie load, where  $2 = \begin{cases} 0.8 : CFRP \\ 0.5 : AFRP \\ 0.3 : GFRP \end{cases}$ a; a; a; recomposition; defection;  $\frac{d\phi}{d\phi}$  cracted rights in the  $E_5$ : The distribution coefficient to calculate the<br> $E_5$ : The distribution coefficient to calculate the<br> $E_5$ I<sub>2</sub>: flexural Highnen in the Uncracted